

Lecture 7

Thermal Insulation & Cryostat Basics

J. G. Weisend II



Goals

- Introduce conduction, convection & radiation heat transfer as they apply to cryogenics
- Describe design techniques to reduce heat transfer into cryogenic devices
- Allow the estimating and scaling of heat leaks into cryogenic devices
- Discuss the basics of cryostat design
- Warning! Not a full description of heat transfer
 - Many topics (boiling, detailed convection calculations, complicated geometries in radiation heat transfer etc) won't be covered
 - Should, however, be a good example of how heat transfer theory can be applied to practical problems.

- Conduction
 - Heat transfer through solid material
- Convection
 - Heat transfer via a moving fluid
 - » Natural or free convection – motion caused by gravity (i.e. density changes)
 - » Forced – motion caused by external force such as a pump
- Radiation
 - Heat transferred by electromagnetic radiation/photons
- There is no such thing as a perfect insulator – though we can design systems with very small heat leaks
- All matter above 0 K radiate heat
 - Remember we can't get to 0 K – 3rd Law of Thermodynamics though we can get vanishingly close
- Heat flows from high temperature to low
 - Heat leaks in, cold doesn't leak out

- Fundamental Equation – The Fourier Law in one dimension

$$Q = -K(T)A(x)\frac{\partial T}{\partial x}$$

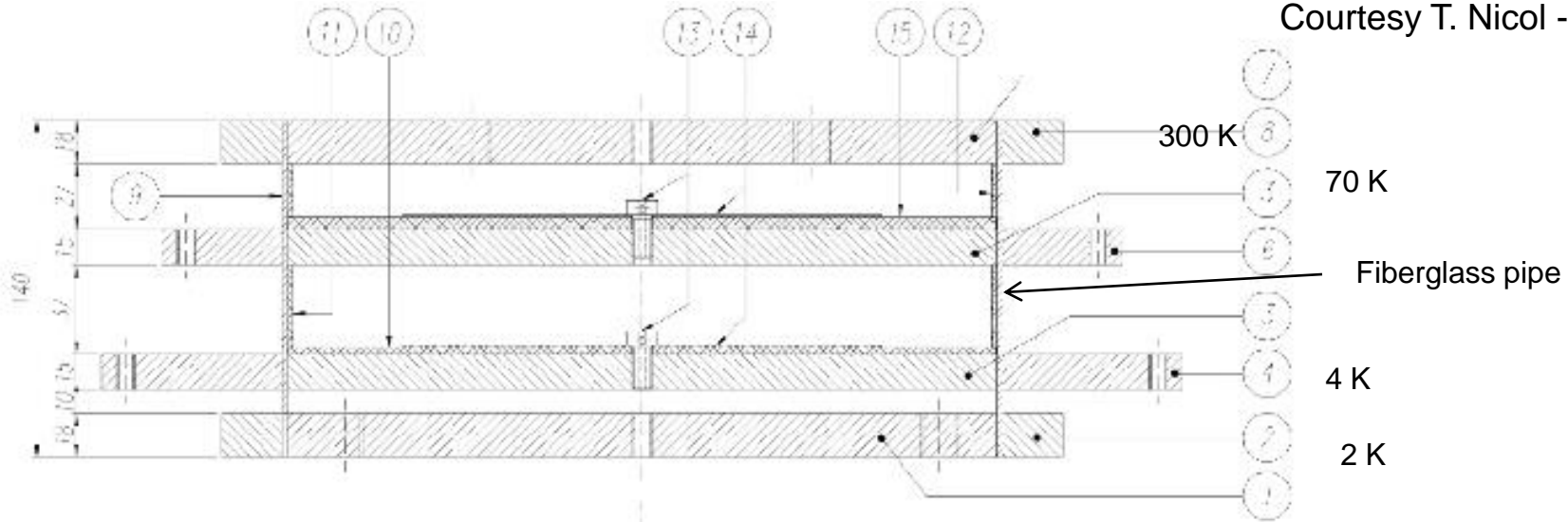
- If we assume constant cross section we get: $Q = -A/L \int_{T_C}^{T_H} K(T)dT$

- Reduce conduction heat leak by:
 - Low conductivity material: make $K(T)$ small
 - Reduce cross sectional area: make A small
 - Increase length: make L large
 - For a given T_C make T_H smaller: i.e. use intermediate temperature heat intercepts
 - » You still have heat leak from 300 K to this intermediate temperature but remember Carnot, It's more thermodynamically efficient to remove heat at higher temperatures



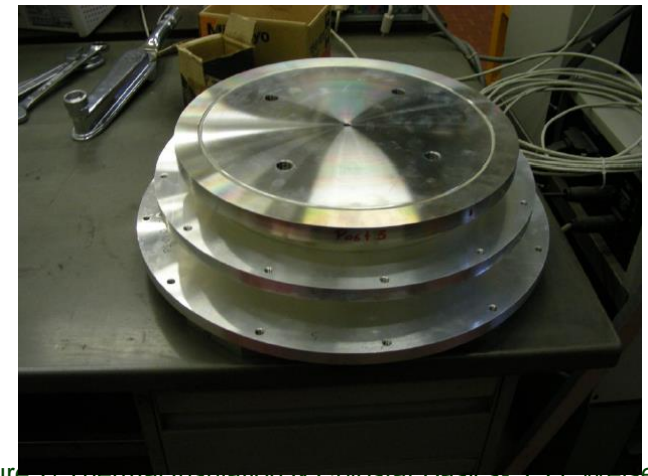
LUND
UNIVERSITY

Design Example ILC Cryomodule Support Post



Courtesy T. Nicol - Fermilab

- Total Heat Leak (conduction & radiation)
 - 70 K - 10.5 W
 - 5 K - 0.9 W
 - 2 K - 0.03 W
- Can support up to 50 kN





Conduction Heat Transfer

- Conduction heat leaks may be estimated by the use of Thermal Conductivity Integrals (Lecture 4)

$$Q = -G(\theta_1 - \theta_2)$$

- Fundamental Equation: Newton's law of cooling

$$Q = hA(T_{\text{surface}} - T_{\text{fluid}})$$

where h is the heat transfer coefficient and is a function of Re , Pr , geometry etc depending on the situation

- In cryogenics we eliminate convection heat leak in cryogenic systems by “simply” eliminating the fluid – vacuum insulation
- Using vacuum insulation to create vessels capable of storing cryogenic liquids was first done by James Dewar – who liquefied hydrogen
 - Such vessels are frequently called dewars – though not always, more later
 - Thermos bottles are a simple example of this approach

Design Example

Vacuum Insulated Test Cryostat



- Contains 3 Vacuum Spaces
 - 1 between 300 K wall and LN₂ bath
 - 1 between LN₂ bath and LHe bath
 - 1 between LHe bath and experiment

- How much vacuum is enough?
 - This of course depends on the heat leak requirements but generally we want to be below 10^{-5} torr. If we maintain this level or better we can generally neglect the convection heat leak for most applications.
 - » Cryogenic Engineering, Flynn (1997) has a good discussion of calculating heat leak due to residual gas pressure
- Cryopumping
 - At cryogenic temperatures almost all common gases condense and freeze onto the cold surface. Typically, we'll see that once surfaces are cooled to ~ 77 K the isolation vacuum will drop to the 10^{-8} torr or better range if the system is leak tight and doesn't have significant outgassing
 - But don't just start cooling with everything at room pressure
 - » Heat leak will likely be too high
 - » Safety hazards due to enrichment of LOX on cold surfaces
 - » Large amounts of condensed gases in vacuum space can lead to other problems including rapid pressure rise upon warming and possible solid conduction
 - » Best practice is to be at least 10^{-3} torr before cooling, lower pressures are better but there may be operational tradeoffs

- All material outgas into a vacuum. This can raise the pressure in a sealed vacuum space
- Reduce outgassing by:
 - Minimize amount of polymers, wire insulation, FRP etc – difficult
 - Keep vacuum surfaces as clean as possible. Remove any oil or cutting fluid, wear gloves etc.
- Getters: materials inserted into vacuum spaces to remove residual gas at low pressures
- In cryogenic systems, getters may be useful in removing residual gas and passively managing small leaks

- 3 types of getters
 - Adsorbers – gas bonds to surface
 - » Activated charcoal, silica gel
 - » Effectiveness increases with decreasing temperature – good for cryogenic systems
 - Chemical getters – chemical reaction between material and gas
 - » Ba & other Alkali metals – not very common in cryogenics
 - Solution or absorber getters – gas is absorbed in interstitial space of metals
 - » Ti, Zr, Th works well with H₂, O₂ and N₂
 - » Much better at room temperature
 - » Occasional use in room temperature applications in cryogenic systems

Getters, Cryogenics and Gilligan's island

- It turns out that one of the most common and effective materials used for getters of low pressure He gas is activated charcoal made from coconut husks.
- There is a significant amount of this material in the LHC magnet cryostats

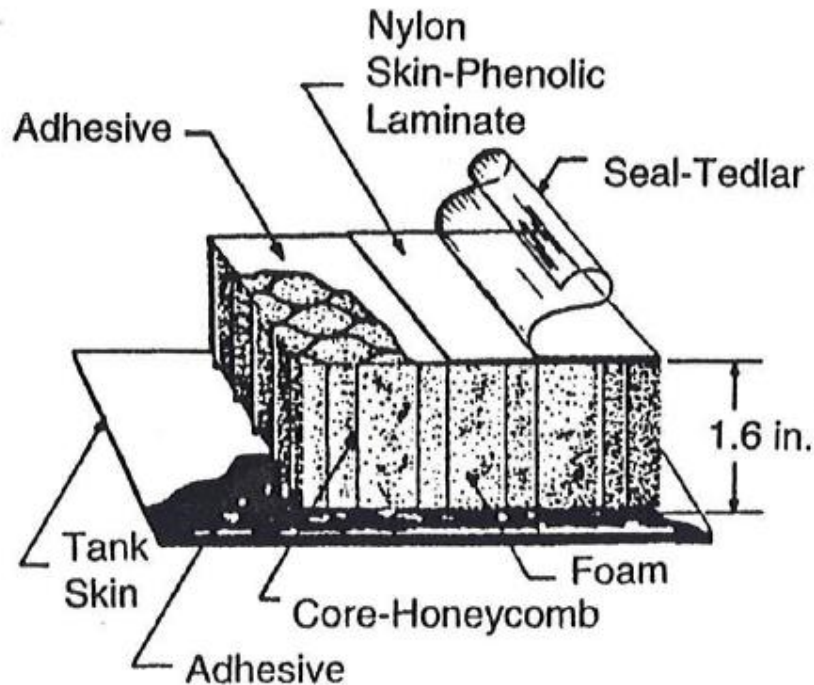


The Professor says:
“Lets look for the Higgs!”

- Not all cryogenic systems use vacuum insulation
- This is particularly true of storage vessels for fluids other than helium
- Reasons for using alternatives to vacuum insulation
 - Cost
 - Weight – Space shuttle main tank
 - Required hold time – related to size
 - Complex vessel shapes
- Some solutions
 - Expanded closed or open cell foams
 - Rock wool, fiberglass or other porous material
- These all require vapor barriers to prevent air from being pulled into the insulation and condensed (can cause both a safety hazard via O₂ enrichment & reduce effectiveness)



Design Example: Complex Foam Insulation System: LH₂ Tank for 2nd Stage Saturn V



From Cryogenic Engineering, Flynn

- Allows helium purging of the insulation
- Weight ~ 4.15 kg/m²
- Performance: measured effective thermal conductivity (0.86 – 1.1 mW/cm K) at $T_{av} = 144$ K Note this includes conduction, convection and radiation heat transfer

- Frequently the largest source of heat leak to cryogenic systems
- Fundamental Equation: Stefan-Boltzmann Law – energy emitted from an ideal black body: $E_b = \sigma T^4$ where $\sigma = 5.67 \times 10^{-8} \text{ W/m}^2\text{K}^4$
- Real world Assumptions:
 - » Emissivity (ε) $\ll 1$ and independent of wavelength (grey body)
 - » Two parallel infinite plates: Radiative heat flux (W/m^2)

Eq. A
$$q_r = \left(\frac{\varepsilon_1 \varepsilon_2}{\varepsilon_1 + \varepsilon_2 - \varepsilon_1 \varepsilon_2} \right) \sigma (T_1^4 - T_2^4)$$

» Frequently in cryogenic systems $\varepsilon_1 \sim \varepsilon_2 \ll 1$ then Eq. A becomes:

Eq. B
$$q_r = \left(\frac{\varepsilon}{2} \right) \sigma (T_1^4 - T_2^4)$$

- » Two long concentric cylinders or concentric spheres (1 represents the inner cylinder): the radiative heat flux (W/m^2) on the inner cylinder is

Eq. C

$$q_1 = \left(\frac{\sigma(T_1^4 - T_2^4)}{\frac{1}{\varepsilon_1} + \left(\frac{A_1}{A_2}\right)\left(\frac{1}{\varepsilon_2} - 1\right)} \right)$$

- » Note as is frequently the case in cryogenics, if the spacing between the cylinders is small compared to the inner radius (i.e. $A_1 \sim A_2$) Eq. C becomes Eq. A

Radiation Heat Transfer

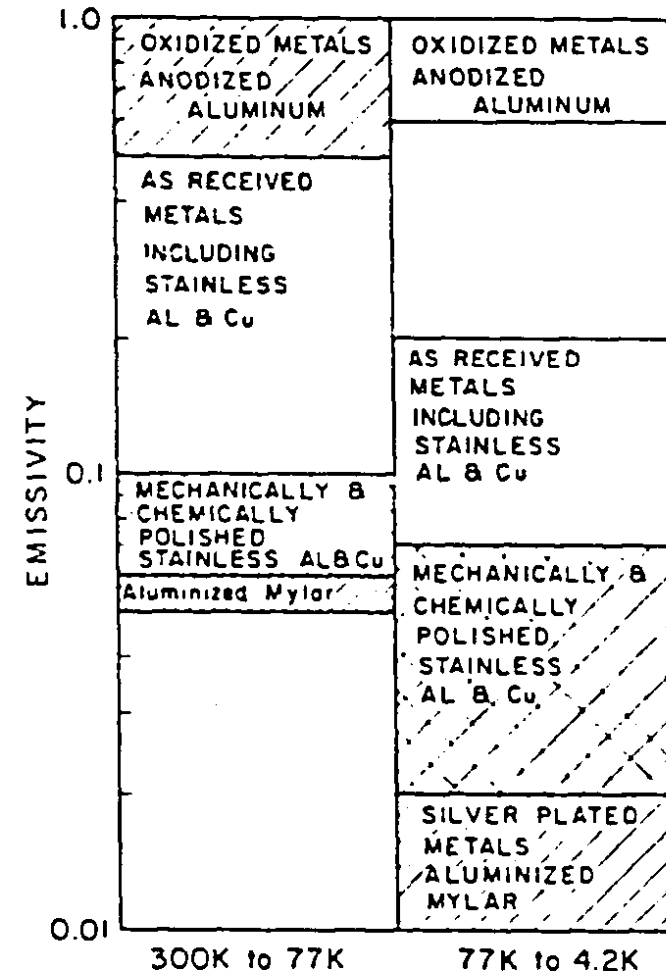


LUND
UNIVERSITY



- Looking at Eq. A, How do we reduce the radiation heat transfer?
- We could reduce the emissivity (ϵ)
 - This is done in some cases; using either reflective tape or silver plating
 - Better below 77 K
 - It's also part of MLI systems (see below)
 - We have to consider tarnishing
 - May be labor intensive

From Helium Cryogenics – S. W. Van Sciver





Radiation Heat Transfer

- Another way to reduce radiation heat transfer is to install intermediate actively cooled radiation shields that operate at a temperature between 300 K and the lowest cryogenic temperature. This has several advantages.
 - It greatly reduces the heat load to the lowest temperature level
 - » Assume parallel plates with $\varepsilon = 0.2$
 - » then from Eq. B $q (300 \text{ K} - 4.2 \text{ K}) = 46 \text{ W/m}^2$ while $q (77 - 4.2) = 0.2 \text{ W/m}^2$
 - It allows heat interception at higher temperatures & thus better Carnot efficiency
 - Such an actively cooled shield provides a convenient heat intercept for supports, wires etc to reduce conduction heat leak.
- Shields may be cooled by
 - Liquid baths (LN_2)
 - Vapor boil off from stored liquid – common in LHe storage dewars
 - Cooling flows from refrigeration plants
 - Conductive cooling via small cryocoolers

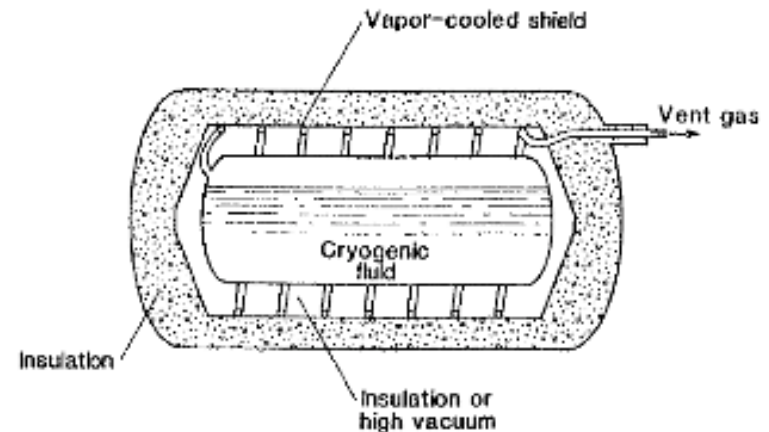
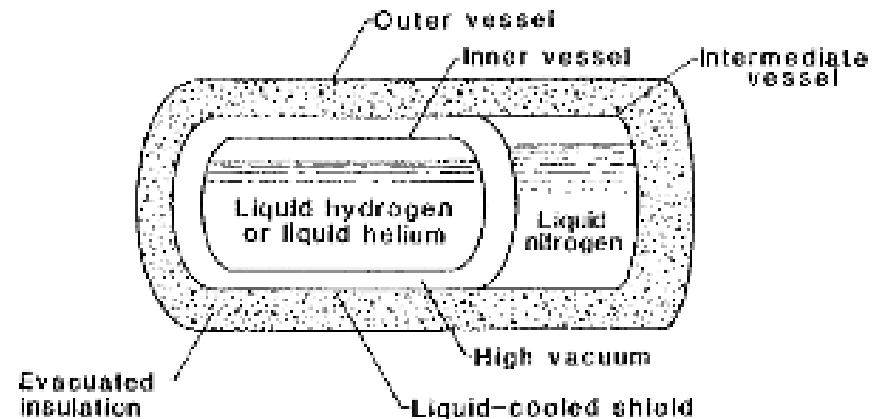
Examples of Cooled Radiation Shields



LUND
UNIVERSITY



- LN₂ bath surrounds inner LHe or LH₂ bath
- Baths are separated by a vacuum insulation space
- Shield is cooled by boil off gas from stored cryogen
 - Spacing of cooling tubes on shield may be calculated by: $\Delta T = qL^2/2kt$
 - » ΔT = max allowable temperature difference between any point on shield and tube
 - » q = heat flux on shield
 - » k = shield thermal conductivity
 - » L = $\frac{1}{2}$ max tube spacing
 - » t = shield thickness



From Cryogenic Engineering, Flynn

Thermal Radiation Shields



LUND
UNIVERSITY

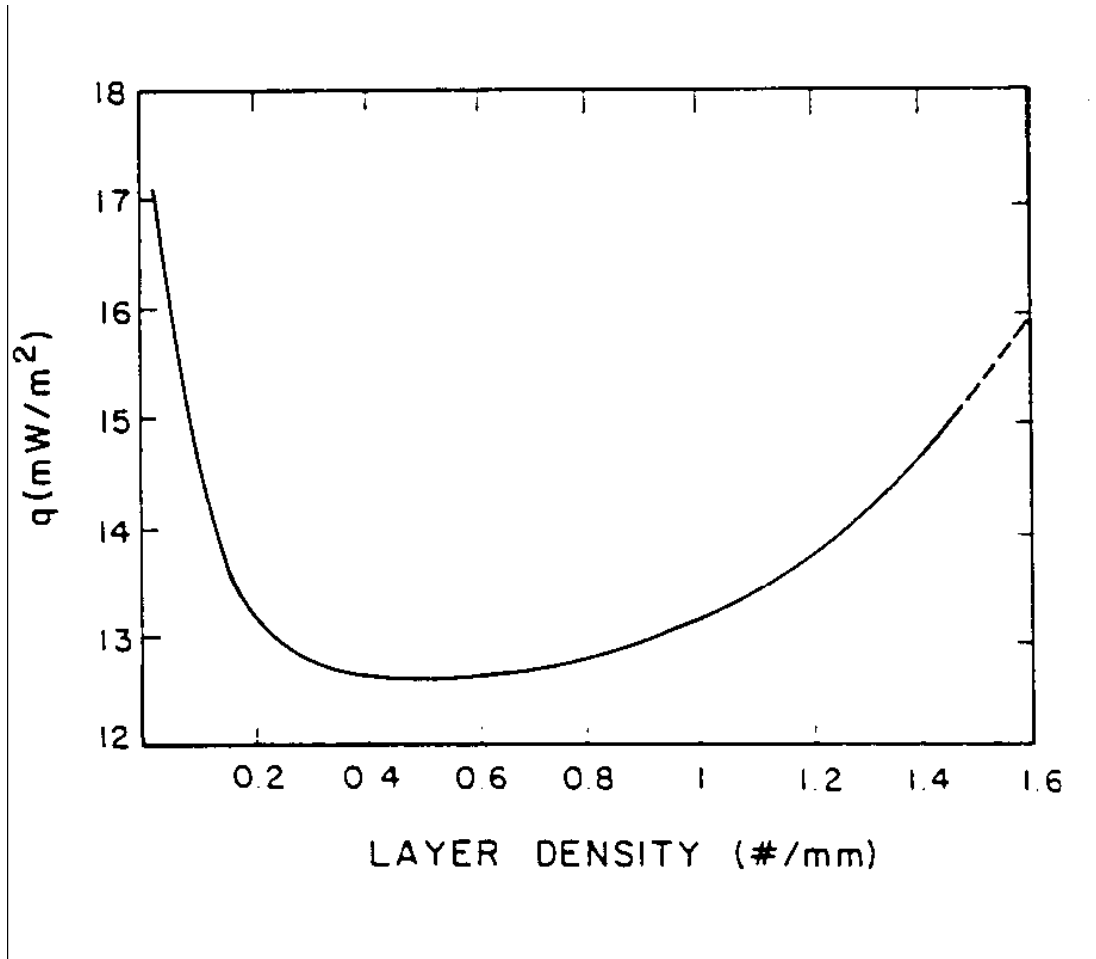


- Uncooled thermal radiation shields placed in a vacuum space between the warm & cold surfaces also help reduce the thermal radiation heat leak
- It can be shown (with the grey approximation and equal emissivities) that with N shields thermal radiation heat transfer is given by:

$$q = \frac{\varepsilon}{(N + 1)2} \sigma (T_H^4 - T_L^4)$$

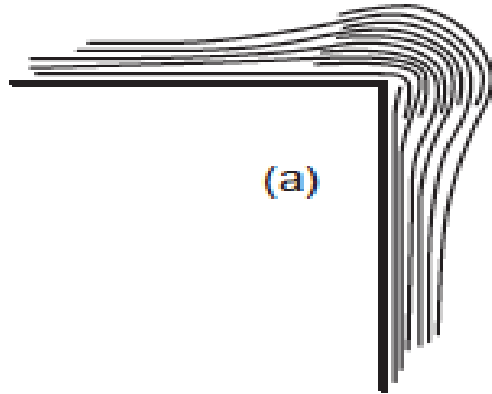
This is the motivation behind Multilayer Insulation

- Also referred to as superinsulation
- Used in the vacuum space of many cryostats (10^{-5} torr or better for best performance)
- Consists of highly reflective thin sheets with poor thermal contact between sheets.
 - Made of aluminized Mylar (or less frequently Kapton)
 - May include separate non conducting mesh
 - May use Mylar aluminized on only one side and crinkled to allow only point contacts between sheets
 - Frequently perforated to allow for better pumping
- Can be made up into blankets for ease of installation
- Don't pack MLI too tightly. Optimal value is ~ 20 layers / inch
- Great care must be taken with seams, penetrations and ends.
 - Problems with these can dominate the heat leak





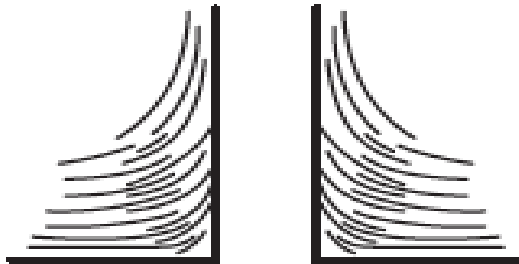
Examples of Proper MLI Installation



(a)



(b)



(c)



(d)

From "Cryogenic Engineering" in *Wiley Mechanical Engineer's Handbook*



LUND
UNIVERSITY

MLI Example from LHC cryostats



“SERIES-PRODUCED HELIUM II CRYOSTATS
FOR THE LHC MAGNETS: TECHNICAL CHOICES,
INDUSTRIALISATION, COSTS”

A. Poncet and V. Parma

Adv. Cryo. Engr. Vol 53

- Radiation heat transfer may also be reduced by filling the vacuum space between 300 K and cryogenic temperatures with other materials that are low conductivity and block line of sight
- Such materials include:
 - Glass beads or microspheres
 - Perlite powder (made from a volcanic rock)
 - Opaciated powders – copper or other metallic flakes mixed in with other powders to further reduce radiant heat transfer
 - Aerogel
- Advantages:
 - Cheaper
 - Easier to install in complex shapes
 - Better performance than MLI in poor or no vacuum
- Frequently used in large storage and transport dewars

The total heat transfer through porous insulation between 2 spheres may be estimated by:

$$W = \frac{\bar{k}(T_2 - T_1)}{t} \sqrt{A_1 A_2}$$

- Where
 - t = thickness of Insulation
 - \bar{k} = the mean thermal conductivity
 - 1 = inner vessel and 2 = outer vessel
- Mean thermal conductivities may be found in references such as Cryogenic Engineering by Flynn

Comparison of Thermal Insulation Approaches (6 inch thick insulation in all cases)

Type of Insulation	Total Heat Flux (W/m ²)	
	300 K to 77 K	77 K to 20 K
Polystyrene Foam (2 lb/ft ³)	48.3	5.6
Gas Filled Perlite powder (5 – 6 lb/ft ³ filled with He)	184.3	21.8
Perlite powder in vacuum (5 – 6 lb/ft ³)	1.6	0.07
High Vacuum (10 ⁻⁶ torr $\epsilon = 0.02$)	9	0.04
Opacified powder (Cu flakes in Santocel)	0.3	-
MLI	0.03	0.007



Increasing Cost & Complexity

Note better performance of evacuated Perlite over high vacuum between 300 K & 77 K



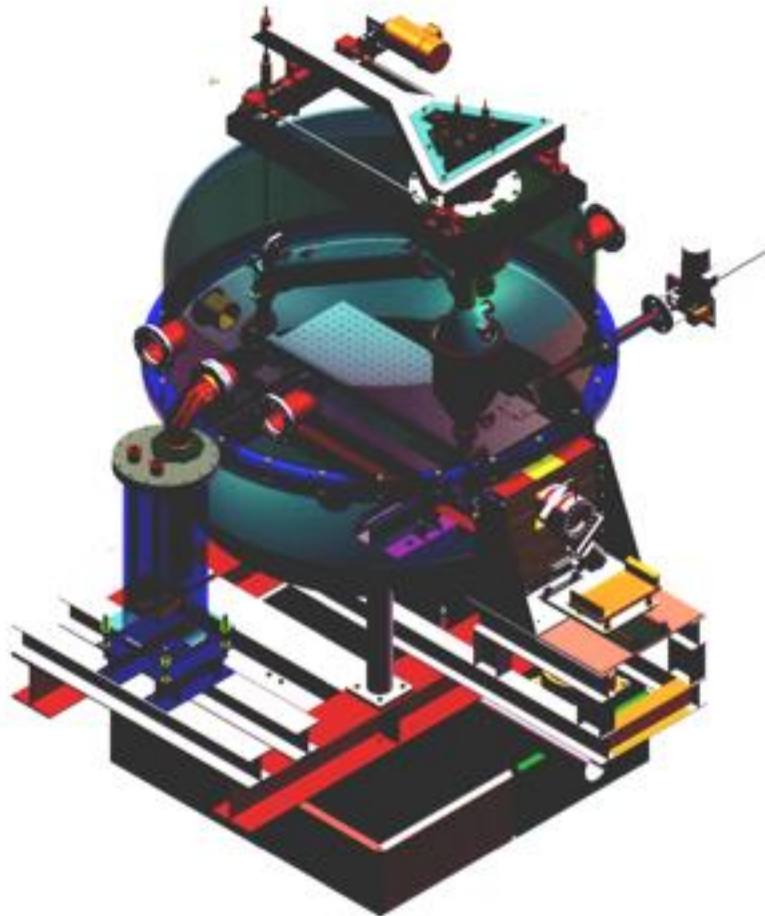
Cryostat Design

- What is a cryostat?
 - A device or system for maintaining objects at cryogenic temperatures.
- Cryostats which contain superconducting RF systems are traditionally called cryomodules (term originally coined by Jlab)
- Cryostats whose principal function is to store cryogenic fluids are frequently called Dewars. Named after the inventor of the vacuum flask and the first person to liquefy hydrogen



LUND
UNIVERSITY

E158 LH₂ Target Cryostat



- Cryostats are one of the technical building blocks of cryogenics
- Cryostat design involves many subtopics:
 - Development of requirements – covered here
 - Materials selection – already covered
 - Thermal insulation - already covered
 - Support systems – covered here
 - Safety – covered in a future lecture
 - Instrumentation – covered in a future lecture
- One of the best ways to learn about cryostat design is through examples (see next 2 talks) Also:
 - Cryostat Design J.G. Weisend II (Ed) Springer (2016)
- There are many different types of cryostats with differing requirements
 - The basic principles of cryostat design remain the same
 - Before we can do anything else we have to define our requirements

- Maximum allowable heat leak at various temperature levels
 - This may be driven by the number of cryostats to be built as well as by the impact of large dynamic heat loads (SCRF or target cryostats)
- Alignment and vibration requirements
 - Impact of thermal cycles
 - Need to adjust alignment when cold or under vacuum?
 - Alignment tolerances can be quite tight (TESLA : +/- 0.5 mm for cavities and +/- 0.3 mm for SC magnets)
- Number of feed throughs for power, instrumentation, cryogenic fluid flows, external manipulators

- Safety requirements (relief valves/burst discs)
 - Design safety in from the start. Not as an add on
- Size and weight
 - Particularly important in space systems
- Instrumentation required
 - Difference between prototype and mass production
- Ease of access to cryostat components
- Existing code requirements (e.g. TUV or ASME)
- Need, if any, for optical windows
- Presence of ionizing radiation

- Expected cryostat life time
- Will this be a one of a kind device or something to be mass produced?
- Schedule and Cost
 - This should be considered from the beginning

All Design is Compromise